Surname	Centre Number	Candidate Number
Other Names		2



GCE A level

1325/01



PHYSICS – PH5 Electromagnetism, Nuclei & Options

A.M. THURSDAY, 18 June 2015

1 hour 45 minutes

ADDITIONAL MATERIALS

In addition to this paper, you will require a calculator, a **Case Study Booklet** and a **Data Booklet**.

INSTRUCTIONS TO CANDIDATES

Use black ink or black ball-point pen. Do not use pencil or gel pen. Do not use correction fluid.

Write your name, centre number and candidate number in the spaces at the top of this page.

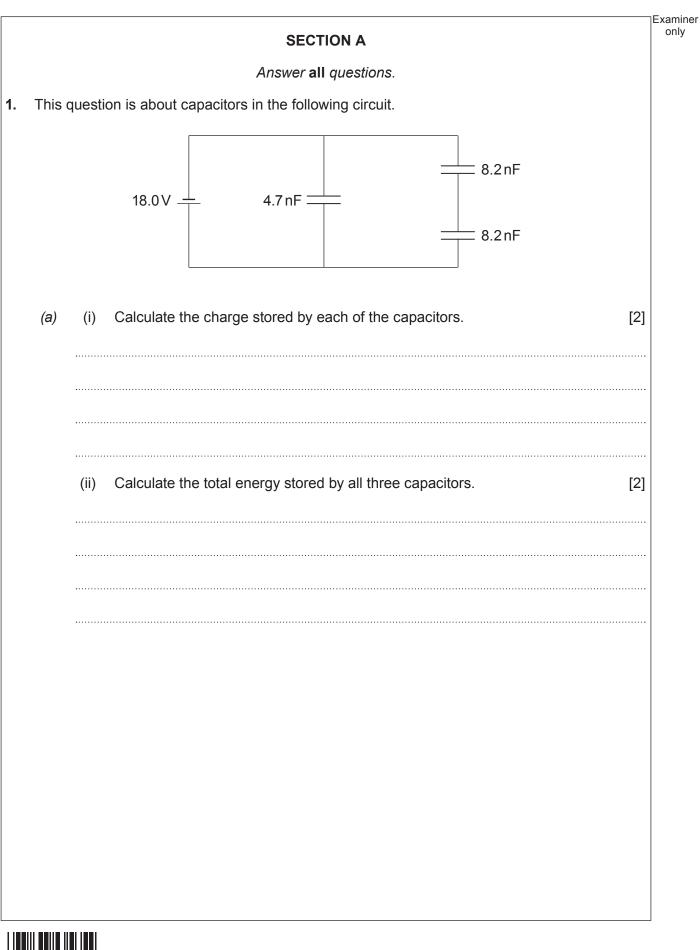
Write your answers in the spaces provided in this booklet. If you run out of space, use the continuation pages at the back of the booklet, taking care to number the question(s) correctly.

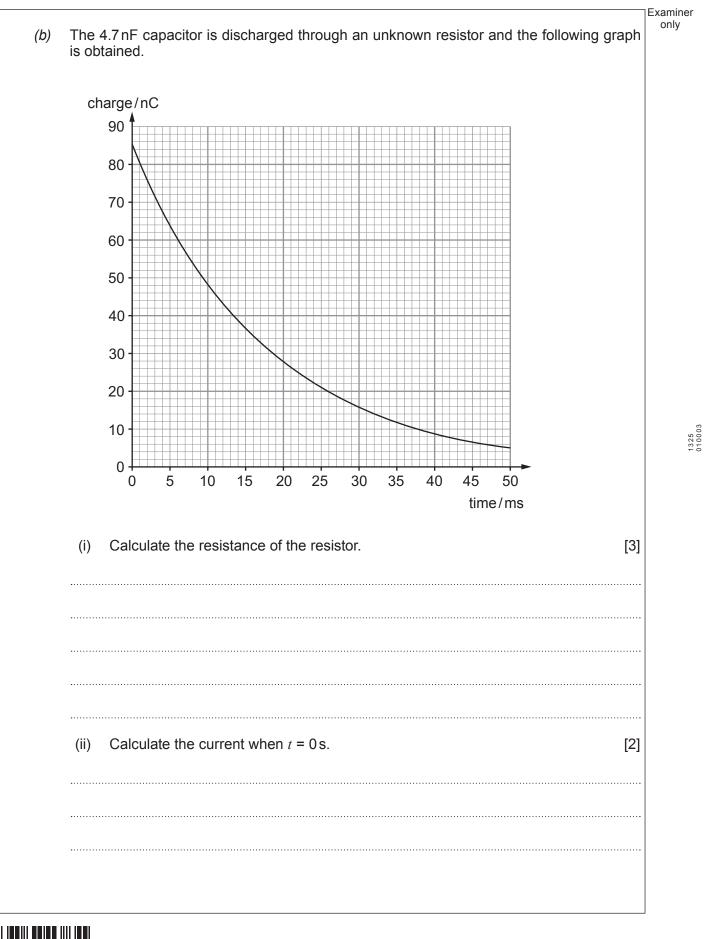
INFORMATION FOR CANDIDATES

This paper is in 3 sections, **A**, **B**, and **C**.

- Section A: 60 marks. Answer **all** questions. You are advised to spend about 1 hour on this section.
- Section B: 20 marks. The Case Study. Answer **all** questions. You are advised to spend about 20 minutes on this section.
- Section C: Options; 20 marks. Answer **one option only.** You are advised to spend about 20 minutes on this section.







(iii)	By estimating the time taken for the capacitor to lose 90% of its charge or otherwise calculate the time taken for the capacitor to lose 99% of its charge. [2]
•••••	
••••••	

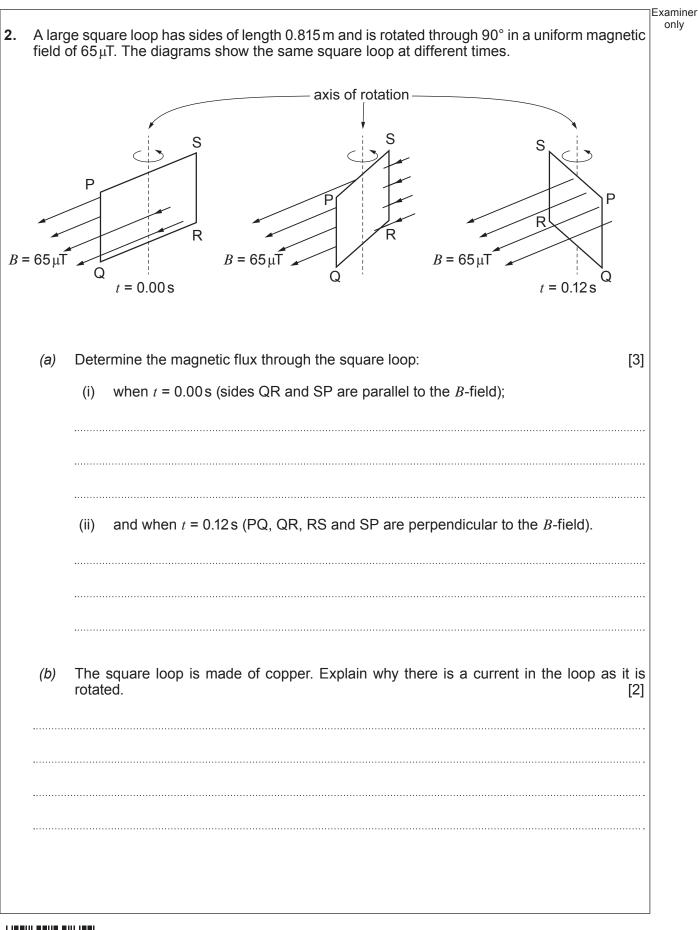




ON THIS PAGE

BLANK PAGE

PLEASE DO NOT WRITE





1325 010007

Examiner only Explain how Lenz's law will give the direction of the forces acting on the sides PQ and RS (C) as the square loop is rotated. [2] The copper wire from which the square loop is made has a circular cross-section of diameter 6.0 mm. The resistivity of copper is $1.67 \times 10^{-8} \Omega$ m. Calculate the **mean** current flowing through the square loop as it is rotated between *t* = 0.00 s and *t* = 0.12 s. [5] (d) copper wire in the shape of a square loop 0.815 m 6.0 mm



only

8 Examiner Some nuclei undergo fusion while others undergo fission. Both processes can result 3. (a) in the release of energy. Discuss these processes in terms of energy and stability. The binding energy per nucleon graph is provided to assist your answer. [4] binding energy per nucleon (MeV/nucleon) ¹⁰⁶₄₈Cd 62 45 Ni Ca $|_{8}^{16}$ O 28 9 12 20 9 ²⁰⁹₈₃Bi ²³⁴₉₀Th $_{26}^{56}$ Fe 8 8 He 7 7 ²³⁸₉₂U 6 6 ²²⁷₈₉Ac ° 5 5 4 4 3 3 $^{3}_{2}$ He 2 2 Η 1 1 20 40 100 120 140 160 180 200 220 240 60 80 0 0 nucleon number



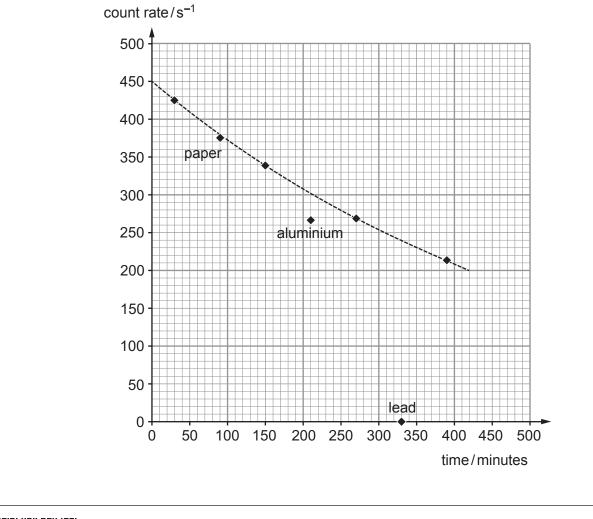
群尧咨询

		9	F i
	(b)	Use data from the graph to estimate the energy released in the reaction. ${}^1_0n + {}^1_1p \longrightarrow {}^2_1H$	[2] Examiner only
	(C)	Use the graph to estimate the energy released in the following reaction. (Hint: use binding energies on both sides of the reaction equation.) $\frac{1}{0}n + \frac{235}{92}U \longrightarrow \frac{137}{55}Cs + \frac{96}{37}Rb + 3\frac{1}{0}n$	the [4]
		0 ¹¹ · <u>92</u> · <u>55</u> · <u>3</u> /R0 · <u>50</u>	
-			
	09	© WJEC CBAC Ltd. (1325-01) Turn ov	/er.

4. Technetium-99 emits **only** γ (gamma) radiation. An experiment was carried out to show this. Various absorbers were placed between the source and detector at the times shown in the table below and the mean count rate was obtained.

Absorber	Time from the start of the experiment/min	Count rate/s ⁻¹
none	30	425
1 sheet (0.1 mm) of paper	90	374
none	150	338
3 mm of aluminium	210	267
none	270	268
10 cm of lead	330	1
none	390	213

These results were plotted on a graph. The decay curve of the technetium-99 itself is plotted as a dotted line which shows the activity dropping continuously as the experiment proceeded.

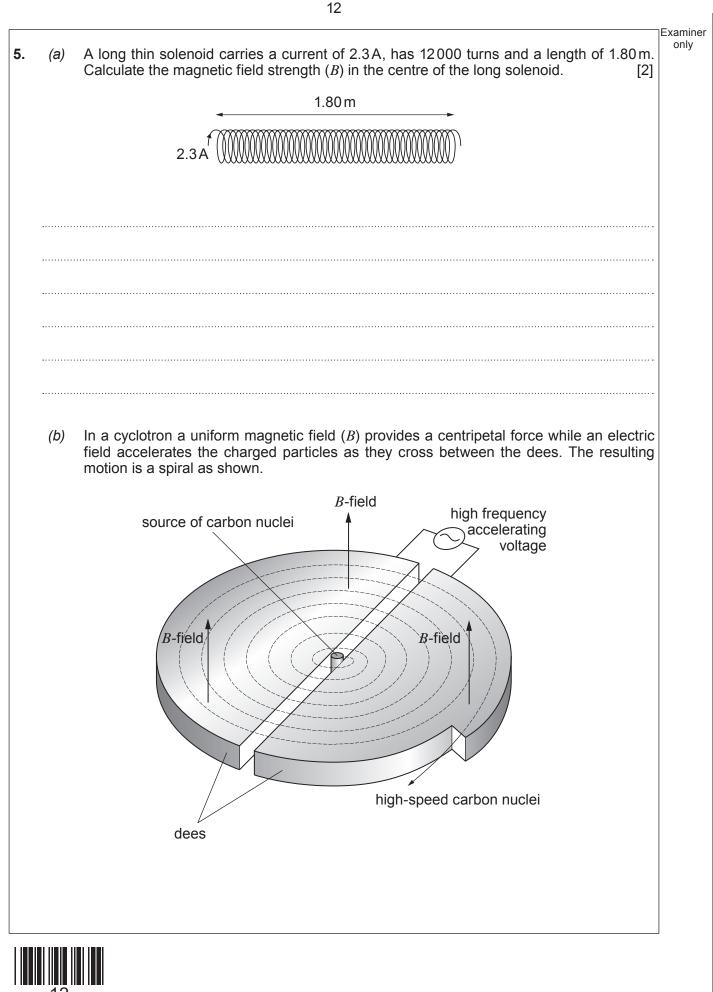




Examiner only

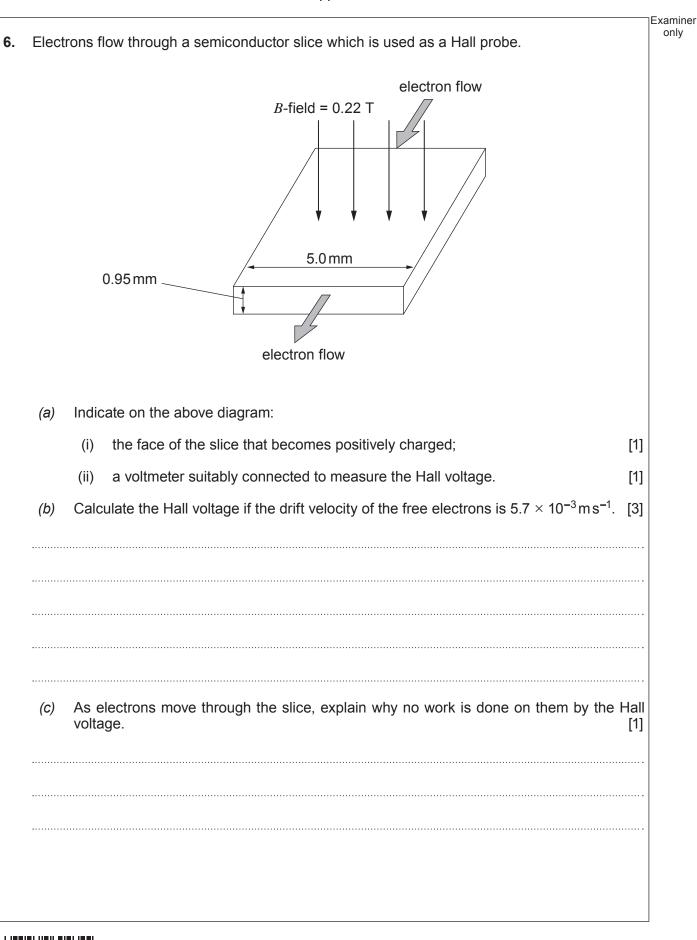
Examiner only Determine the half-life of technetium-99. (a) [1] Explain why the results are consistent with technetium-99 only emitting γ radiation. (b) [3] The detector only detects 0.6% of the γ radiation emitted by the source. Use the graph and the half-life of technetium-99 to calculate the initial mass of technetium-99 (the mass (C) of a technetium-99 atom is 99 u). [4]





cyclotron and the potential difference between the dees is 14.5 kV (assume that the	וכא ס [3]	By equating the centripetal force to the magnetic force, show that the frequence the a.c. supply is given by: $f = \frac{Bq}{2\pi m}$
 <i>m</i> = 12 u in a strong <i>B</i>-field of 3.3 T. (iii) Calculate the final speed of a carbon nucleus after it has completed 12 'orbits' of the cyclotron and the potential difference between the dees is 14.5 kV (assume that the final the dees is 14.5 kV (assume that the dees is		
 <i>m</i> = 12 u in a strong <i>B</i>-field of 3.3 T. (iii) Calculate the final speed of a carbon nucleus after it has completed 12 'orbits' of the cyclotron and the potential difference between the dees is 14.5 kV (assume that the final speed of a carbon nucleus after it has completed 12 'orbits' of the cyclotron and the potential difference between the dees is 14.5 kV (assume that the final speed of a carbon nucleus after it has completed 12 'orbits' of the cyclotron and the potential difference between the dees is 14.5 kV (assume that the final speed of a carbon nucleus after it has completed 12 'orbits' of the cyclotron and the potential difference between the dees is 14.5 kV (assume that the final speed of a carbon nucleus after it has completed 12 'orbits' of the cyclotron and the potential difference between the dees is 14.5 kV (assume that the cyclotron and the potential difference between the dees is 14.5 kV (assume that the cyclotron and the potential difference between the dees is 14.5 kV (assume that the cyclotron and the potential difference between the dees is 14.5 kV (assume that the cyclotron and the potential difference between the dees is 14.5 kV (assume that the cyclotron and the potential difference between the dees is 14.5 kV (assume that the cyclotron and the potential difference between the dees is 14.5 kV (assume that the cyclotron and the potential difference between the dees is 14.5 kV (assume that the cyclotron and the potential difference between the dees is 14.5 kV (assume that the cyclotron and the potential difference between the dees is 14.5 kV (assume that the cyclotron and the potential difference between the dees is 14.5 kV (assume that the cyclotron and the potential difference between the cyclotron and the pot		
cyclotron and the potential difference between the dees is 14.5 kV (assume that the		Calculate the cyclotron frequency for a carbon nucleus with $q = 6e$ and n $m = 12u$ in a strong <i>B</i> -field of 3.3 T.
cyclotron and the potential difference between the dees is 14.5 kV (assume that the		
		cyclotron and the potential difference between the dees is 14.5 kV (assume that







Examiner only The concentration of free electrons in the semiconductor slice is $7.0\times10^{22}\,m^{-3}.$ (d) Calculate the current in the slice. [2] _____

			Exar
		SECTION B	
		Answer all questions.	
		The questions refer to the case study. Direct quotes from the original passage will not be awarded marks.	
(a	a)	Give two reasons why only a small fraction of the work done in compressing the gas is transferred to gravitational potential energy of the football (paragraphs 3 & 4). Note that losses due to heat and sound are negligible. [2]	
 (b)	Use the values $u = 20 \text{ m s}^{-1}$, $m_0 = 1.5 \text{ kg}$ and $\frac{\Delta m}{\Delta t} = 5.9 \text{ kg s}^{-1}$ to calculate the speed of the rocket after 0.175 s (paragraph 11 and equation 2). [2]	
 (C	c)	Check that the units (or dimensions) of equation 4 are correct. $\frac{\Delta m}{\Delta t} = \pi r^2 \rho u$ [2]	
••••••			
.			
·····			
·····			
······			
	6		

Examiner only (d) Calculate the exhaust speed of water from the rocket assuming a rate of change of mass of 9.5 kg s^{-1} and the radius of the bottle neck is 1.1 cm using equation 4 (density of water = 1000 kg m^{-3}). [2] Using your own words explain why 'the actual rocket does not keep up with its theoretical (e) counterpart' (paragraphs 16-19 and equation 6). [3] (f) Calculate the initial exhaust speed of water leaving a bottle pumped to a pressure of 7.8×10^5 Pa (the outside atmospheric pressure is 1.0×10^5 Pa) using equation 6 (density of water = 1000 kg m^{-3}). [2]



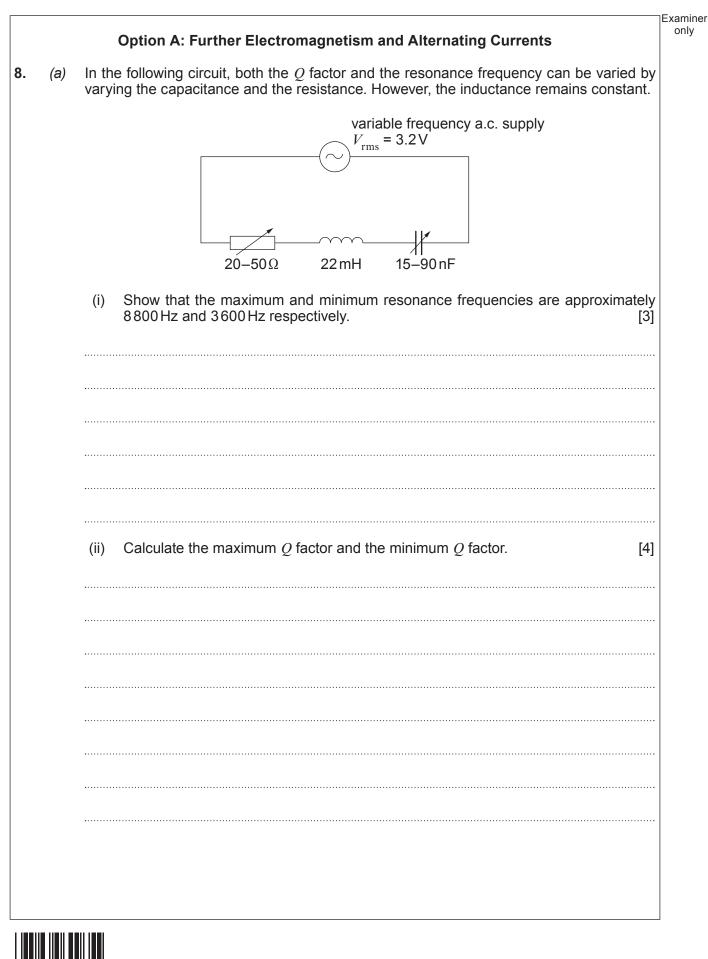
18

Term Description During the first 0.2 s, this term $\pi r^2 \rho u^2$ Thrust force from exhaust increases constant decreases $\pi r^2 \rho u^2$ Thrust force from exhaust / / / ng		Complete the	table, the first row has been c	ompleted for	you (paragr	aphs 20-22).	[3]
$\frac{ \mathbf{r} ^2}{ \mathbf{r} ^2} \frac{ \mathbf{r} ^2}{ $		T	Description	During th	ne first 0.2s,	this term	
$\frac{\pi r^{\mu}\rho u^{\mu}}{mg} \qquad \qquad$		Ierm	Description	increases		decreases	
$0.0107v^2$ $0.0107v^2$ n) Show that the first term $(\pi r^2 \rho u^2)$ in equation 8 can be written as $2(p - p_{atm}) \times A_{neck}$ where A_{neck} is the cross-sectional area of the bottle opening (see equation 5 or 6). [2] <i>ii</i>) In practice, using Boyle's law is inappropriate because the gas cools as it expands. [1] (ii) Explain why little or no heat flows when the gas in the bottle expands. [1] (iii) Use the first law of thermodynamics to explain why the temperature of the gas		$\pi r^2 \rho u^2$				1	
 <i>i</i>) Show that the first term (πr²ρu²) in equation 8 can be written as 2(p - p_{atm}) × A_{neck} where A_{neck} is the cross-sectional area of the bottle opening (see equation 5 or 6). [2] <i>i</i>) In practice, using Boyle's law is inappropriate because the gas cools as it expands. (i) Explain why little or no heat flows when the gas in the bottle expands. [1] (ii) Use the first law of thermodynamics to explain why the temperature of the gas 		mg					
 <i>A</i>_{neck} is the cross-sectional area of the bottle opening (see equation 5 or 6). [2] (i) In practice, using Boyle's law is inappropriate because the gas cools as it expands. (i) Explain why little or no heat flows when the gas in the bottle expands. [1] (ii) Use the first law of thermodynamics to explain why the temperature of the gas 		$0.0107v^2$					
				e opening (se	ee equation	5 or 6).	[2]
	(1)	A _{neck} is the cro	oss-sectional area of the bottle	te because th	ne gas cools	as it expands	

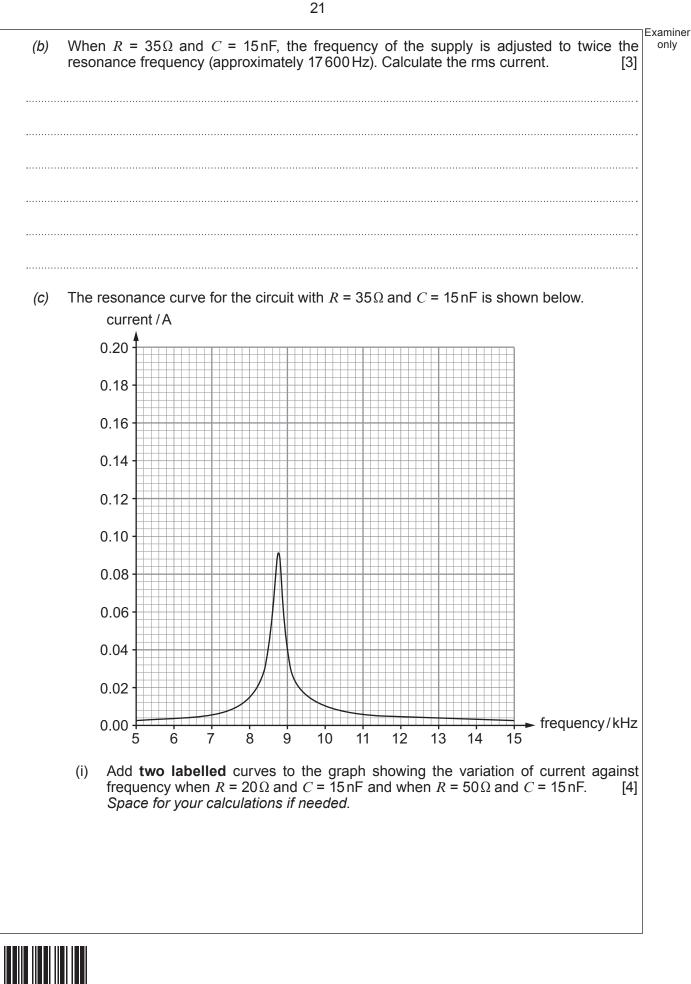


	SECTION C: OPTIONAL TOPICS	
Option A:	Further Electromagnetism and Alternating Currents	
Option B:	Revolutions in Physics – The Newtonian Revolution	
Option C:	Materials	
Option D:	Biological Measurement and Medical Imaging	
Option E:	Energy Matters	
Answer the	e question on one topic only.	
Place a tic	k (\mathcal{I}) in one of the boxes above, to show which topic you are answering.	
You are a	dvised to spend about 20 minutes on this section.	









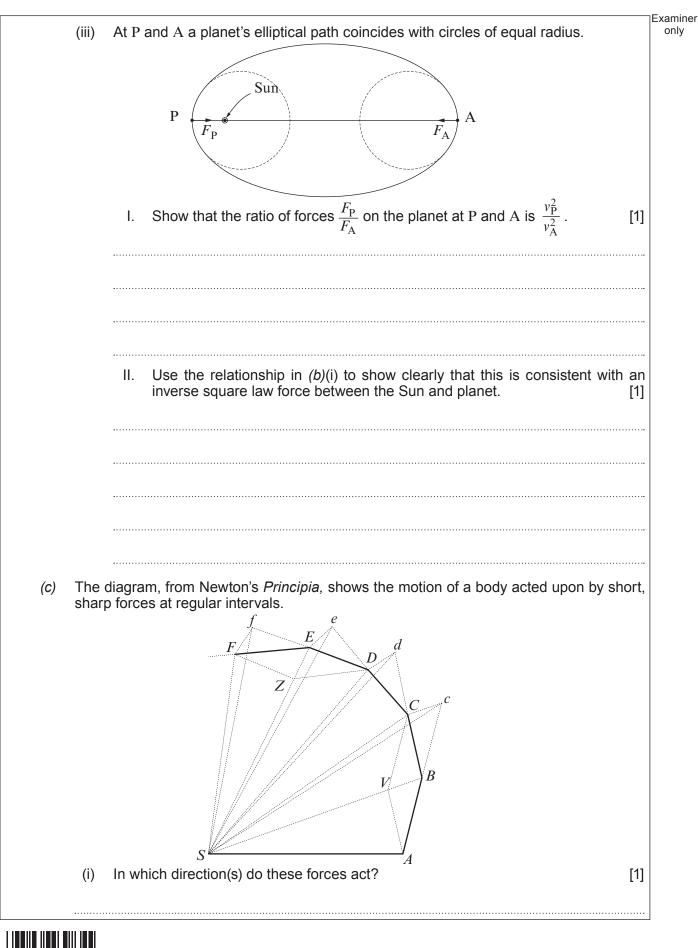
(i	i) Use the equation $Z = \sqrt{\left(\omega L - \frac{1}{\omega C}\right)^2 + R^2}$ to explain in detail why the current varies	Exa o
(.	i) Use the equation $Z = \sqrt{\left(\omega L - \frac{1}{\omega C}\right)^2 + R^2}$ to explain in detail why the current varies with frequency as shown in the graph (no calculations are required). [6]	5]
·····		
·····		

			Option B: Revolutions in Physics – The Newtonian Revolution	Exai oi
(8	a)	(i)	Tycho Brahe and another astronomer, 570 km away, were able to make simultaneous angle measurements on heavenly bodies. They could measure parallax down to about $\frac{1}{60}$ degree (0.017°) due to their different locations. Show clearly, giving a labelled diagram , that the furthest distance away of a body for which they could measure parallax was about 2 × 10 ⁶ km. Assume the body to be directly overhead. [3]	
		(ii)	Tycho Brahe and his associate could easily measure parallax for the Moon $(0.40 \times 10^{6}$ km away) but were barely able to detect the parallax of a comet which appeared in 1577. Explain why this provided evidence against Aristotle's division of the universe into sublunary and superlunary (beyond the Moon) regions, where different laws applied. [2]	



Examiner only A planet's speed and distance from the Sun at perihelion (P) and aphelion (A) are (b) (i) related by: $r_{\rm P} v_{\rm P} = r_{\rm A} v_{\rm A}$ Derive this from Kepler's equal area law (Kepler's second law), adding to the diagram to assist your explanation. [3] Sun Р A For Jupiter, $r_{\rm P}$ = 7.41 \times 10⁸ km and $r_{\rm A}$ = 8.16 \times 10⁸ km. Evaluate the **percentage** (ii) change in Jupiter's speed as it goes from A to P. [2]





Examiner only (ii) State what Newton was able to show about the triangles ASB, BSC, CSD and how the argument could be applied to the motion of planets. [3] (iii) Over forty years before Newton published the Principia, Descartes had proposed a quite different explanation for why planets orbited the Sun. What was Descartes' explanation, why did many people find it more satisfying than Newton's, and why is it now almost forgotten? [4]

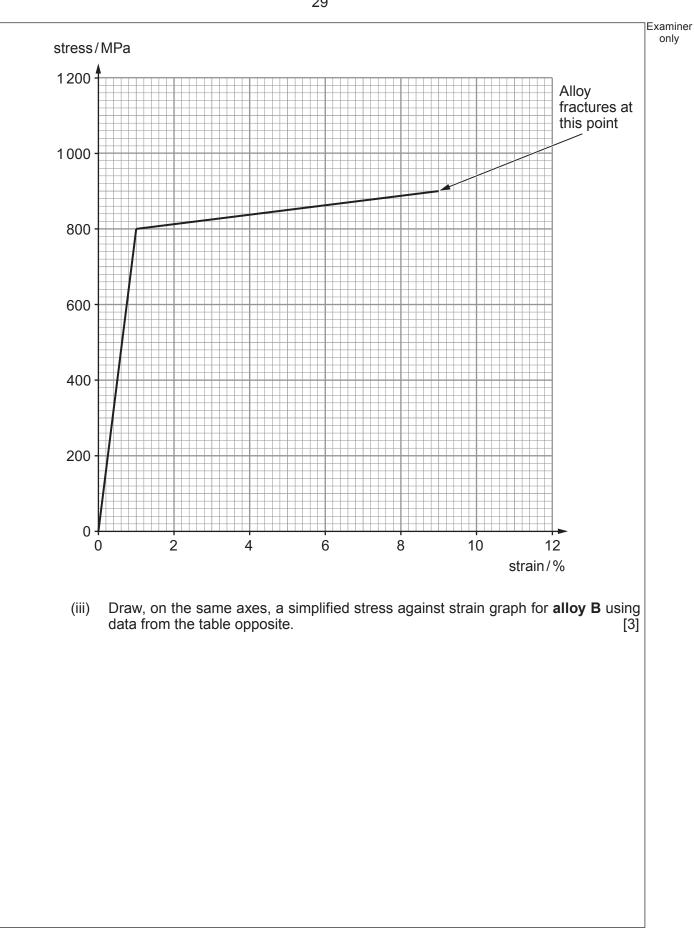
Examiner only **Option C: Materials** 10. The picture shows the microscopic structure of glass, an amorphous solid. (a) 00 Si Explain the following macroscopic properties of glass. Glass fibres are brittle, showing no plastic deformation before fracture. [2] (i) A sheet of glass can be fractured accurately and cleanly if its surface is scratched (ii) and then the glass is bent slightly. [You may wish to draw a labelled diagram to support your answer.] [2]

only

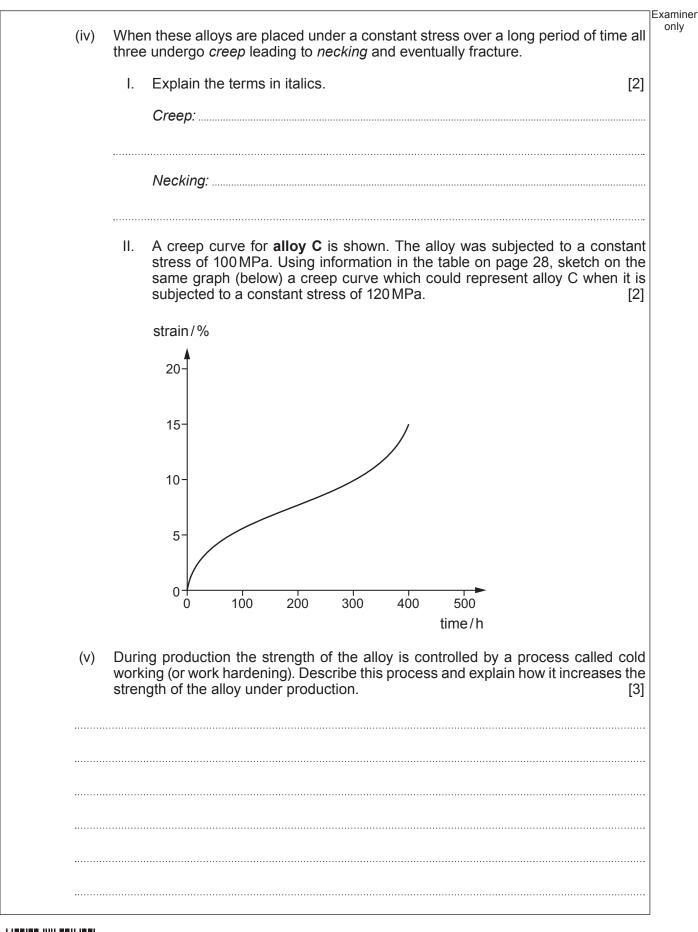
28

Examiner (iii) During production, car windscreens are strengthened by using jets of cold air to cool the outer surfaces of the hot glass. This causes the outside to contract quickly while the inside remains soft. Later, the inside cools and contracts. Explain how this process makes the windscreen difficult to break. [2] Advanced aircraft and spacecraft applications require aluminium based alloys which are (b) able to operate under extremely diverse conditions. The properties of three of these alloys are given in the table. Aluminium alloy Young modulus/GPa Yield strength/MPa Maximum tensile strain/% Alloy A 9 80 800 Alloy B 80 1000 5 Alloy C 60 600 15 The graph opposite shows the stress against strain graph for **alloy A**. Use information from the graph to confirm that the Young modulus of alloy A is (i) 80 GPa. [1] Alloy A is in the form of a cylinder of length 2.5 m and diameter 2.5 mm. Determine (ii) the work done to stretch this alloy to breaking point. [3]

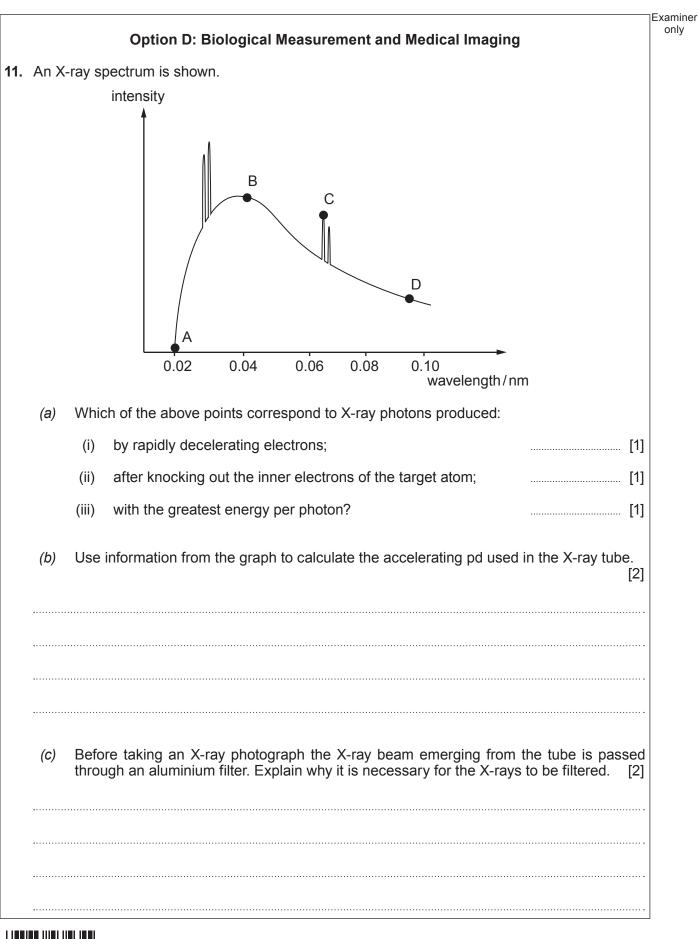








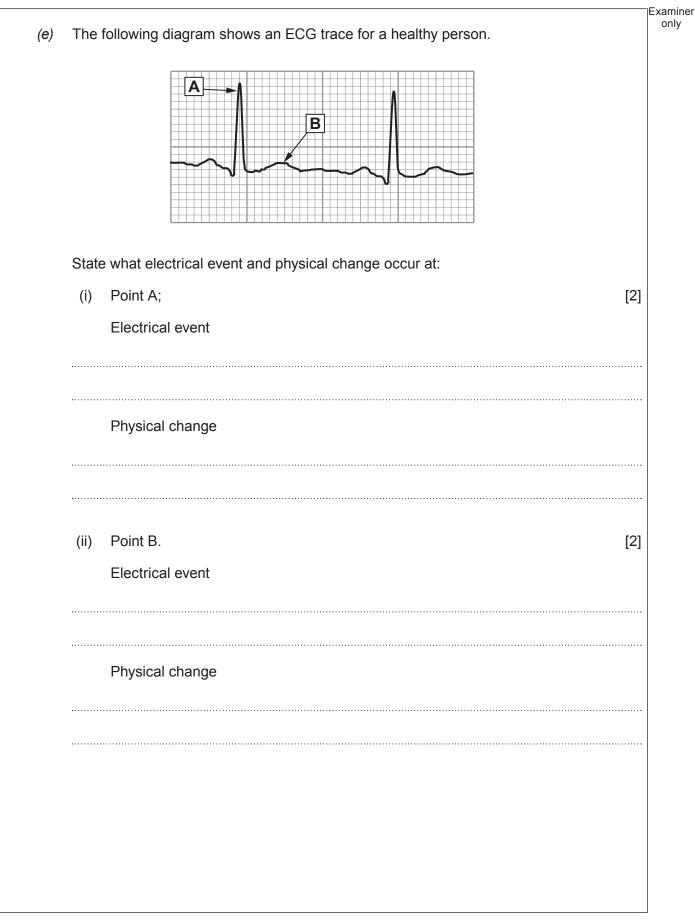




© WJEC CBAC Ltd.

 (ii) Briefly explain the function of this electromagnetic radiation in the working of an MRI scan. [2] (iii) Describe one disadvantage of MRI scanning as an imaging technique. [1] 	MRI scan. [2]	d)	(i)	Name the region of the electromagnetic spectrum used in MRI scans.	[1]
(iii) Describe one disadvantage of MRI scanning as an imaging technique. [1]	(iii) Describe one disadvantage of MRI scanning as an imaging technique. [1]		(ii)	Briefly explain the function of this electromagnetic radiation in the working of MRI scan.	
(iii) Describe one disadvantage of MRI scanning as an imaging technique. [1]	(iii) Describe one disadvantage of MRI scanning as an imaging technique. [1]				
			/:::)		
				Describe one disadvantage of MRI scanning as an imaging technique.	







(f)	Ultrasound can be used to measure the speed at which blood is flowing. When reflected off a red blood cell, the wavelength of the ultrasound changes.		
	(i)	What is the name given to this effect? [1	
	(ii)	If ultrasound of wavelength 500 μ m is used, its speed when travelling through blood is 1500 m s^{-1} and the wavelength received at the detector is 500.4μ m. Calculate the speed of the flow of blood.	
(g)	(i)	PET scans are often used to detect tumours. What part of the electromagnetic spectrum do PET scanners detect?	
	(ii)	Why are PET scanners not commonly used in district hospitals? [1	
34			

		Option E: Energy Matters	Exa
12.	(a)	Explain why it is important to enrich uranium before it is suitable to be used in a fission nuclear power station. [3]	
	(b)	In a breeder nuclear reactor uranium-238 is changed into plutonium. Explain the advantage of this and how it is achieved. [2]	
	(c)	State two possible advantages of deuterium-tritium fusion over uranium-235 fission. [2]	



		centration (number per m ³) of deuterium and tritium particles (<i>n</i>). These conditions are ally expressed as: $\tau Tn \ge 3.5 \times 10^{28}$ s K m ⁻³
	(i)	Explain why a high temperature (<i>T</i>) is necessary. [3
	(ii)	A confinement time (τ) of 0.9 s and a temperature of 120 million Kelvin are attainable Calculate the minimum density of plasma required in kgm ⁻³ (the mean mass of deuterium and tritium ions is 2.5 u). [3
	·····	
(e)	The energy that can be produced from 1 kg of uranium-235 is 8.3×10^{13} J whereas the energy available from 1 kg of deuterium-tritium is 3.4×10^{14} J. Calculate the energy that can be produced from 1 kg of anti-matter (remember that anti-matter and matter annihilate).	



Examiner only Coal, biomass, uranium-235, natural gas and wind are five energy resources with similar mean costs per MWh of energy production (\pounds 40- \pounds 60 per MWh). Discuss other advantages (f) and disadvantages of all five energy resources. [5] _____ **END OF PAPER**





群尧咨询

38

BLANK PAGE

PLEASE DO NOT WRITE

ON THIS PAGE

Question	Additional page, if required. Write the question number(s) in the left-hand margin.	Ex
number	Write the question number(s) in the left-hand margin.	
	······	



uestion umber	Additional page, if required. Write the question number(s) in the left-hand margin.	
		••••





GCE A level 1325/01-B PHYSICS – PH5 ASSESSMENT UNIT



A.M. THURSDAY, 18 June 2015

CASE STUDY FOR USE WITH SECTION B

Examination copy To be given out at the start of the examination. The pre-release copy must not be used.

Paragraph

4

2

Rocket Science without the Chemistry

Introduction

It's quite remarkable how much analysis of rocket motion can be done when one is armed with some physics, a bit of mathematics and a spread sheet.

A simple rocket system that converts plastic bottles to rockets can be bought relatively cheaply. A good example is the aquapod[®], upon which the system in the photograph is based. It's an upside-down, pressurised plastic bottle, half filled with water. When released, water is ejected at high speed from its tail end resulting in the rocket accelerating upwards. Because downward momentum is given to the water, the rocket gains upward momentum. Early on in the investigations into these rockets, it was discovered that a ball placed on top of the bottle was far easier to investigate and model.

Energy Analysis Considering Gravity

The simplest possible analysis that can be done to the rocket is to apply conservation of energy. We could approximate that the work done in compressing the gas eventually becomes gravitational potential energy of the ball. How much energy is stored in the compressed gas? For this, we need a bit of A level Maths in order to calculate the area below an isothermal compression in a p-V diagram. This is the result of the integration (that doesn't require learning). 3

$$W = p \times V \times \ln\left(\frac{p}{p_{\text{atm}}}\right)$$
 Equation 1

where p is the high pressure inside the bottle, V is the volume of compressed air inside the bottle and p_{atm} is atmospheric pressure.

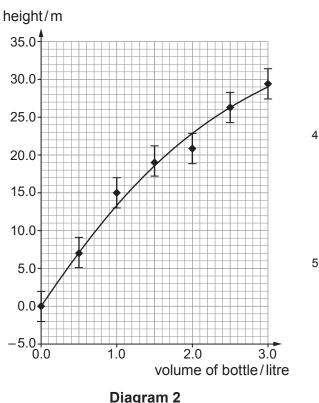
So a 2 litre bottle half filled with water and with a pressure of 4.4×10^5 Pa inside it can supply around 650 J of energy. If all this energy were transferred to a 0.45kg ball, the ball should attain a height of some 150m. In practice, however, only a small fraction of this energy is transferred to the ball.

This experiment was carried out with a set of plastic bottles varying in volume from 500 ml to 3.0 litre. Each bottle was half full of water at take-off and each bottle was pumped to a pressure of 4.4×10^5 Pa (from an initial atmospheric pressure of 1.0×10^5 Pa). The results are shown in the graph (diagram 2).



(1325-01B)





Paragraph

8

325 11B003

Unsurprisingly perhaps, this simplistic conservation of energy argument has not been successful and the relationship between maximum height of the rocket and volume of the bottle is not ⁶ directly proportional as was predicted by conservation of energy.

Ideal Rocket Theory (fixed exhaust speed and ignoring gravity)

Now for some more detailed analysis. We need to look at what causes the acceleration of the rocket. In effect, we need to apply Newton's second law to the rocket and to do that we need to 7 know the rate of change of momentum of water leaving the bottle. First, let's define some terms:

 m_0 = total initial mass of the rocket;

t = time counting from blast-off;

 $\frac{\Delta m}{\Delta t}$ = constant rate of ejecting of mass;

u = constant speed of the water leaving the bottle (relative to the rocket).

The resultant force exerted on the water is equal to its rate of change of momentum. The momentum gained by the water per second is $u\frac{\Delta m}{\Delta t}$ (remember, $\frac{\Delta m}{\Delta t}$ is the mass leaving the bottle per second). This, therefore, is the force experienced by the water and by Newton's 3rd law, this 9 is also the force experienced by the rocket. So, we now know that the thrust force acting on the rocket is $u\frac{\Delta m}{\Delta t}$.

The mass of the rocket is decreasing at a constant rate of $\frac{\Delta m}{\Delta t}$ so its mass at any time *t* is given 10 by $\left(m_0 - \frac{\Delta m}{\Delta t}t\right)$.

This is enough information to use Newton's 2^{nd} law (F = ma) to give an equation for the acceleration. Mathematics (that won't require learning) then leads to solutions for both velocity 11 and height. Here are the equations:

 $v = -u \ln(1 - \alpha t)$ Equation 2 $h = \frac{u}{\alpha} [(1 - \alpha t) \ln(1 - \alpha t) + \alpha t]$ Equation 3

 α is the ratio of the rate of loss of mass to the initial mass, $\left(\alpha = \frac{\Delta m}{\Delta t}{n_0}\right)$.

We can now try to apply these equations to a typical 2 litre bottle. The total mass of the water, bottle and ball is around 1.5 kg (i.e. $m_0 = 1.5$ kg). Of this, 1.0 kg is water, 0.45 kg for the ball and the bottle has a mass of 0.05 kg. From high speed video analysis of the rocket, all 1.0 kg of the vater is expelled in 0.14 s so that we can calculate the mean rate of decrease of mass of the rocket:

$$\frac{\Delta m}{\Delta t} = k = \frac{1.0}{0.14} = 7.1 \,\mathrm{kg \, s^{-1}}$$

Paragraph

13

We can also use these figures to determine the exhaust speed of the water when we know that the radius of the bottle neck is 1.1 cm.

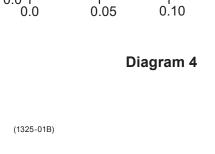
The mass of the water (Δm) in the cylinder of length $u \times \Delta t$ is: volume \times density = $\pi r^2 \times u \times \Delta t \times \rho$ which gives us a relationship between the rate of mass loss and the exhaust velocity: $\frac{\Delta m}{\Delta t} = \pi r^2 \rho u$ Equation 4 When this expression is equated to the actual rate of loss of mass (7.1 kg s⁻¹), it gives a value of u of around 19 m s⁻¹ (remembering that the density of water is rather conveniently 1 000 kg m⁻³). $u \times \Delta t$ Diagram 3

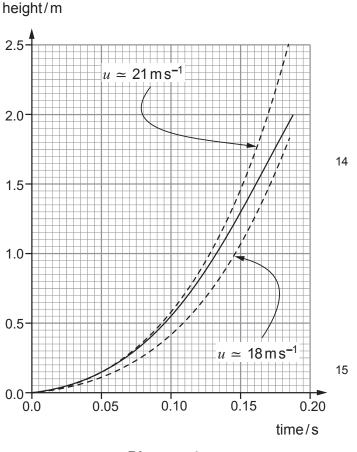
Comparison Between Ideal Rocket Theory and Experiment

This is enough information to put into the rocket equations (2 & 3) and compare with the motion of an actual rocket. The easiest way of doing this is to use a spreadsheet - we can enter both the rocket equations and actual data and compare the two. In the graph shown (diagram 4), the actual 2 litre rocket data is shown as a continuous line. The theoretical rocket equation is represented by the dotted lines. The data for the height of the actual rocket was gathered by using a comparatively cheap digital camera set to 220 frames per second. The video of the rocket motion was then analysed frame by frame using a 30 cm ruler to measure distances on the 1.0 screen and the continuous curve in the graph obtained.

Interestingly, when these data and the 0.5 rocket equations were put into a spreadsheet, the value of exhaust speed (*u*) of 19 ms^{-1} did not produce an ideal fit (see diagram 4). The best fit for the early motion of the rocket was provided by an exhaust speed of around 21 ms^{-1} whereas the later motion of the rocket fits better with an exhaust speed of around 18 ms^{-1} . This may seem like a bad agreement but, on the other hand, these discrepancies could be pointing toward the reason for the disagreement.

© WJEC CBAC I td





Paragraph

Rocket Theory with Decreasing Pressure and Exhaust Speed

The rocket equation fits well for the first 0.10 s of its flight with an exhaust speed of around 21 m s^{-1} but then the actual rocket does not keep up with its theoretical counterpart and the actual rocket seems to fit better with an exhaust speed of around 18 m s^{-1} . What could be the ¹⁶ reason for this? Put simply, it's the decrease in pressure of the air inside the bottle as the water is leaving. But how can we model the pressure inside the bottle? Simplistically, we can use Boyle's law. Robert Boyle in the mid 1600s said for a fixed mass of gas at constant temperature:

pressure × volume = constant

We know that the initial volume of the gas is around 1 litre (for the 2 litre bottle). The final volume of the gas rather obviously will be 2 litre. Boyle's law therefore tells us that the pressure at the start will be approximately double the final pressure. In between these two stages, each gramme ¹⁷ of water that is expelled provides an extra 1 cm³ of air in the bottle and the corresponding pressure drop can easily be calculated using Boyle's law.

Now that we have the details to model the pressure drop in the bottle, it is possible to calculate the speed of the water coming out of the bottle. All we have to do is use Bernoulli's equation.

$$p_{\rm atm} = p - \frac{1}{2}\rho u^2$$
 Equation 5

Surprisingly enough, this means that the exit speed of the water is independent of the size of ¹⁸ the bottle opening!

$$u = \sqrt{\frac{2(p - p_{atm})}{\rho}}$$
 Equation 6

Now we can use this equation to calculate the exhaust speed of the water. The density of water (ρ) is 1000 kg m⁻³ and the initial ($p-p_{atm}$) was 3.4×10^5 . This gives an initial exhaust ¹⁹ speed of around 26 m s⁻¹.

For completion, gravity and air resistance should also be incorporated into our model. Gravity is easy enough but what about air resistance? A simple theory for air resistance is that the increase in air resistance is proportional to velocity squared. In fact, if we look up the air resistance of a sphere, we should find:

$$F_{\rm drag} = 0.47 \times \frac{1}{2} \rho_{\rm air} v^2 \times A$$
 Equation 7

where *A* is the maximum cross-sectional area of the sphere, ρ_{air} is the density of air and *v* is the speed of the sphere.

Another great advantage of placing a football on top of the water bottle rocket is that the air resistance can be modelled based on the dimensions of the football. This assumes that the bottle underneath the football has no effect on the air resistance but should be a reasonable approximation considering that the cross-sectional area of the football is far greater than that of the bottle. The density of air (ρ_{air}) is 1.20 kg m⁻³ and the diameter of the football is 22.0 cm and ²¹ they can both be inputted into the air resistance equation.

height/m

6

Paragraph

22

All this information should give us a final resultant force acting on the rocket of:

$$F_{\rm res} = \pi r^2 \rho u^2 - mg - 0.0107 v^2$$
 Equation 8

where: u = instantaneous exhaust speed of the water

r = radius of the bottle opening

 ρ = density of water (1000 kg m⁻³)

m = instantaneous mass of the rocket (including the water and football)

v = instantaneous speed of the rocket

Final Comparison between Theory and Experiment

When all this data is put into a spreadsheet with time going up in steps of 1/220th of a second ²³ (to match the digital camera) and all rocket data calculated for all the time intervals. This is the end result.

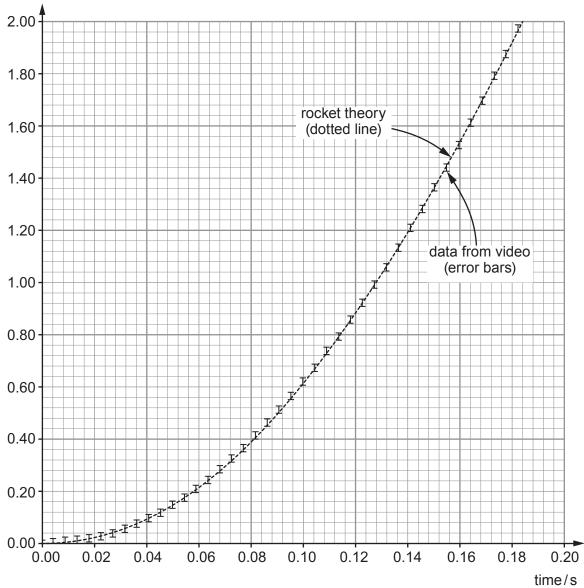


Diagram 5

Paragraph

Here, the rocket theory (dotted line) is in excellent agreement with the experimental results (error bars). At no point is the computed model data outside the error bars corresponding to the actual motion of the rocket (these error bars simply correspond to ± 0.5 mm reading from the ²⁴ ruler next to the computer screen). The final best fit parameters used were $m_0 = 1.52$ kg, initial pressure = 4.7×10^5 Pa and radius of bottle opening = 1.019 cm.

In conclusion, the motion of a plastic water bottle rocket has been analysed using purely A level Physics with a touch of Bernoulli's equation and drag theory. Although the mathematics used can be complicated, this is relatively easily remedied by using numerical methods in a computer ²⁵ spreadsheet. The results are astonishingly accurate and were aided hugely by the novel idea of a football on top of the rocket.